

AIRS PERFORMANCE DURING SPACECRAFT THERMAL VACUUM (TVAC) TESTING

Thomas S. Pagano

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Agenda



- **TVAC Accomplishments and Timeline**
- **Performance Results**
 - *Special Test Results*
- **Summary and Conclusions**



TVAC ACCOMPLISHMENTS AND TIMELINE



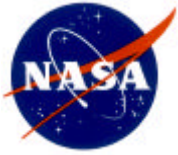
TVAC WAS A MAJOR MILESTONE FOR AIRS



- TRW testing showed no influence from spacecraft or other instruments
- AIRS performed extremely well. No instrument related anomalies detected
- AIRS runs cooler than expected. This means better potentially longer mission life.



11/5/2001



THANKS TO THE TVAC TEST TEAM FOR THE LONG HOURS AND EXCELLENT EFFORT

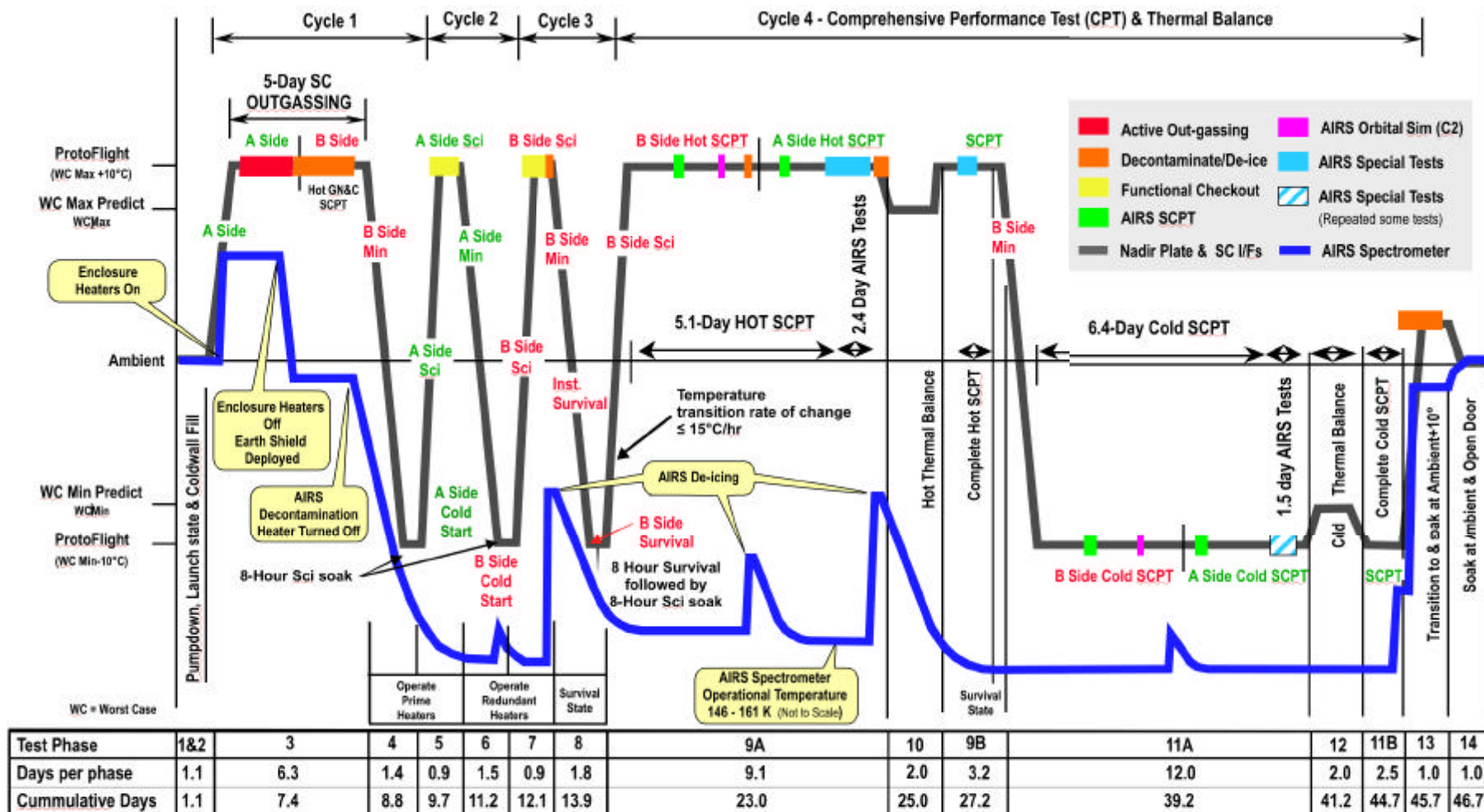


Your efforts have allowed us to demonstrate that we can successfully operate the AIRS instrument in orbit and characterize its performance!

**(SPECIAL THANKS TO THE MANY OTHERS WHO CONTRIBUTED
BUT WERE NOT ABLE TO MAKE THE PHOTO)**



THERMAL VACUUM TESTING AT TRW TOOK ALMOST 47 DAYS





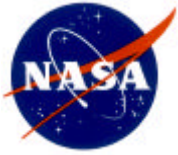
AIRS TVAC ACCOMPLISHMENTS



- ***AIRS instrument and all subsystems performed extremely well***
- ***Accomplishments include:***
 - Earth Shield deployed as commanded
 - Detector dewar vacuum integrity test verified no change from BAE tests
 - Coolers work very efficiently and reliably for long periods of time when left undisturbed.
 - Scanner can be started and operated at a lower temperature
 - Focal plane is fully operational and shows gain ratios equivalent to BAE test data
 - AMA can be commanded to known position and works as expected
 - Spectrometer thermal control can be maintained with high stability
 - Performance sensitivity to thermal state (nominal temp and gradients) characterized
 - The AIRS spectrometer can be maintained at a lower temperature set point
 - AIRS operation procedures work as designed
 - Special Calibration Tests work as designed
- ***CONCERNS / LIENS***
 - On-orbit outgassing plan and timeline needs to be reviewed to prevent ice formation on foreoptics
 - Spacecraft initiated time jams cause major disruptions to the normal operation of AIRS.
 - AIRS on-orbit thermal model must be updated based on the new thermal data set



AIRS INSTRUMENT PERFORMANCE RESULTS



SPECIAL CALIBRATION TEST SEQUENCES (STS) A KEY ELEMENT OF IN-FLIGHT CAL PLAN



- **Transfer pre-flight calibration to in-orbit configuration**
 - Same tests performed pre-flight at TRW and in-orbit
 - Tests are traceable to pre-flight calibration using NIST traceable sources
 - Check location of spectral response functions
 - Re-establish instrument linear radiometric response
- **Discover and quantify potential new sources of stray light and noise**
 - Stray light in the space viewport
 - Determine orbital dependence of noise
 - Set Radiation Circumvention Levels
- **Correct for launch environmental changes**
 - Adjust AMA for AB Balance and Spectral Centering



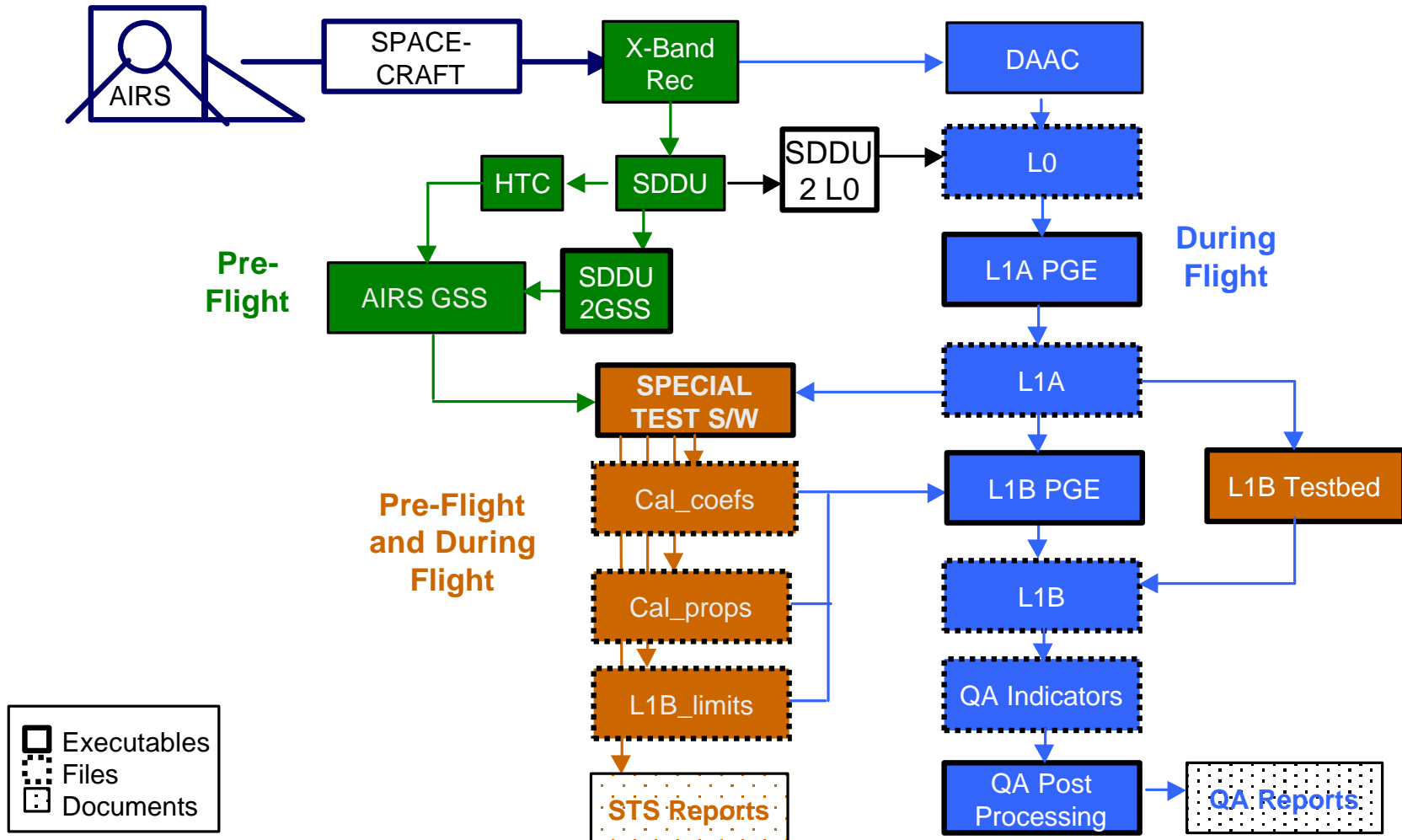
TWELVE SPECIAL TEST OBTAIN KEY MEASUREMENTS



Test ID	Name	Description	Measurement Obtained
AIRS-C1	Normal Mode / Special Events	Establish normal DCR and Lamp operation. Flag data for special events Earth Scene targets of opportunity.	Focal Plane Model Geolocation SST Acquisitions
AIRS-C2	Guard Test	Cycles through A, B and A/B Optimum Gains and acquires data.	Radiometric Gains NEdT Spectral FP Model (Parylene)
AIRS-C3	Channel Spectra Phase	Heat and cool spectrometer by $\pm 1K$	Phase of Channel Spectra
AIRS-C4	AMA Adjust	AMA is moved to the desired x (spatial) and y (spectral) position.	AB Balance Spectral Adjust
AIRS-C5	OBC Cool	Blackbody heater is turned off	IR Linearity
AIRS-C6	Variable Integration Time	Integration time is varied on readout while scanning	Electronics Linearity
AIRS-C7	Space View Noise	The scan mirror is stopped and parked at OBCs	Noise Behavior (Pops, FPN, etc) Drift Characterization
AIRS-C8	Radiation Circumvention	Same test as AIRS-C7 but with radiation circumvention turned on.	Threshold Levels
AIRS-C9	Scan Profile	Slow part of scan rotated to OBCs	Stray Light Calibrator Centration
AIRS-C10	Lamp Operations	Each of the three lamps are exercised by user command.	VIS Gains, VIS Noise
AIRS-C11	Warm Functional	Focal Plane Power is Cycled Test Pattern Gain Table Loaded	FPA Functionality Data Stream Verification
AIRS-C12	Cold Functional	Same as AIRS-C11 except performed cold.	FPA Functionality



MANY PATHS USED TO OBTAIN CAL DATA PRE-FLIGHT AND DURING FLIGHT

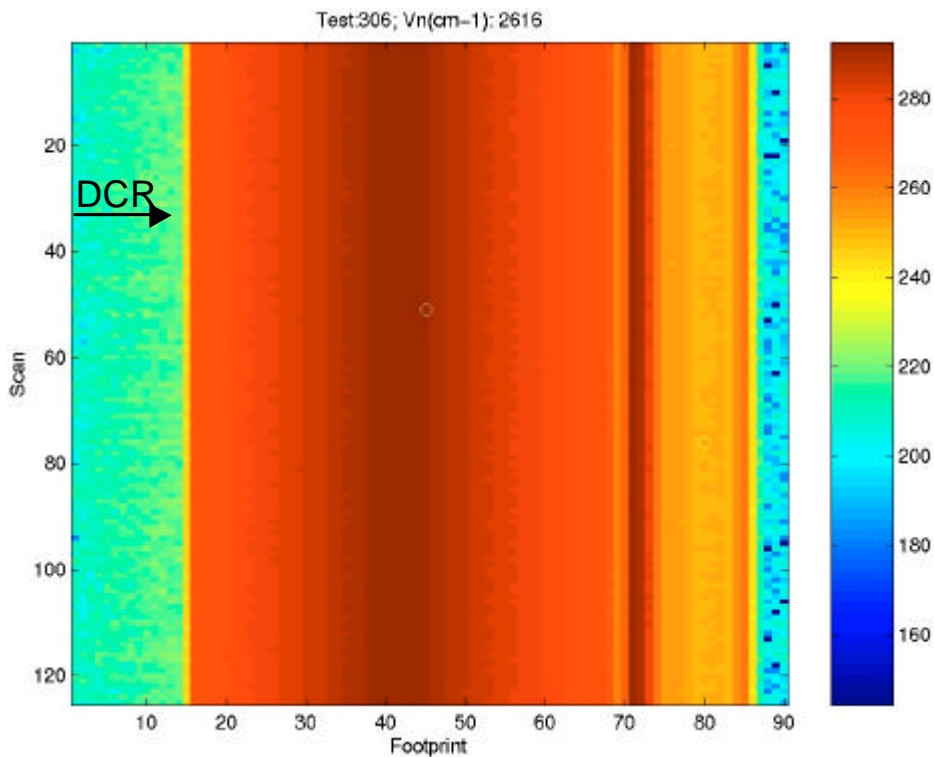




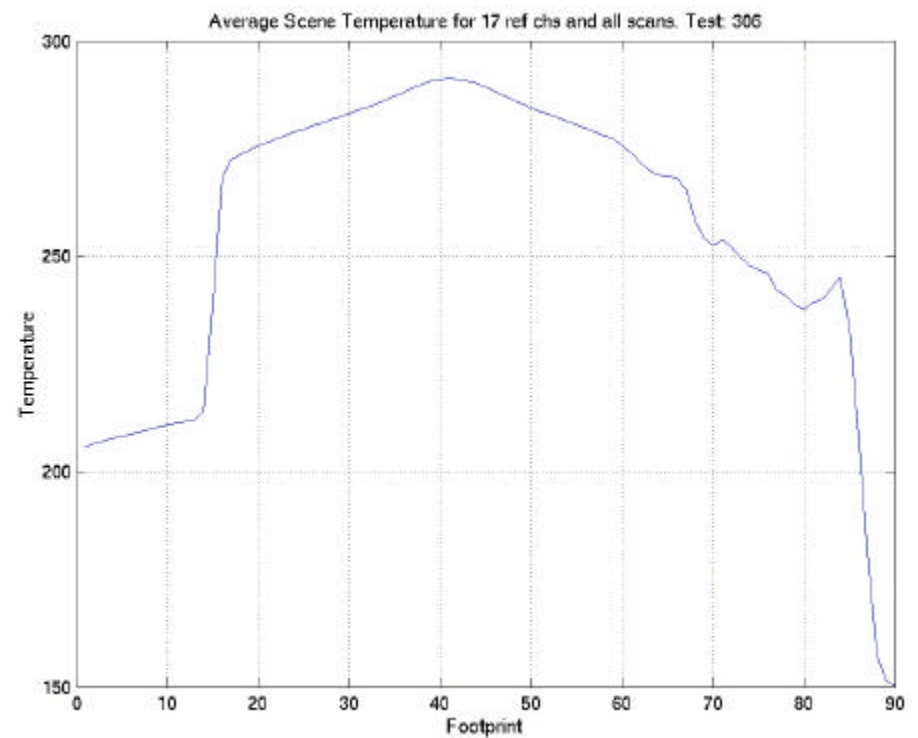
C1: NORMAL MODE IMAGERY OF NADIR PANEL LOOKS GOOD



AIRS THERMAL IMAGE*



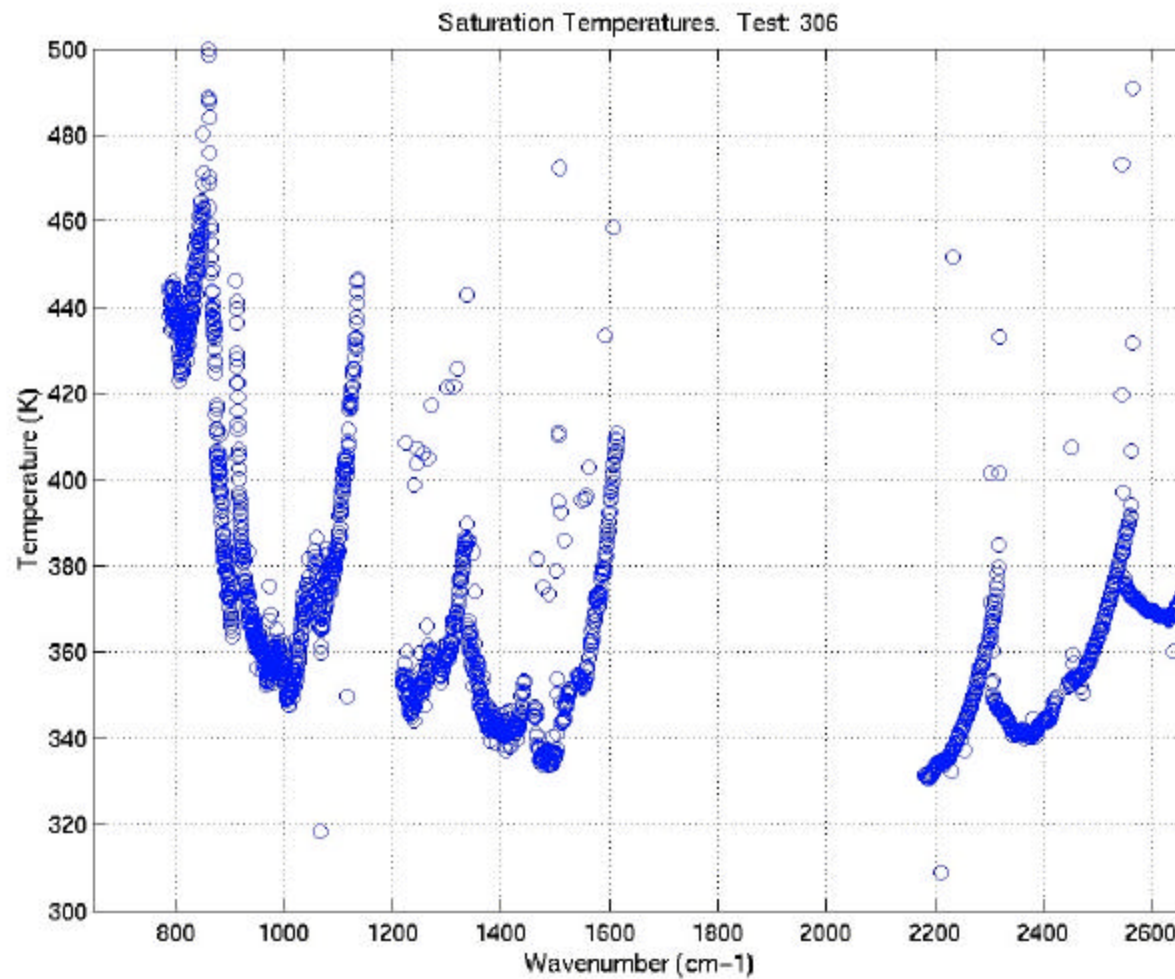
NADIR PANEL TEMPERATURE



- C1 TEST USED TO EXPEDITE DATA AND INITIATE DCR AND PERIODIC LAMP OPERATIONS
- TVAC VERIFIED PROPER OPERATION



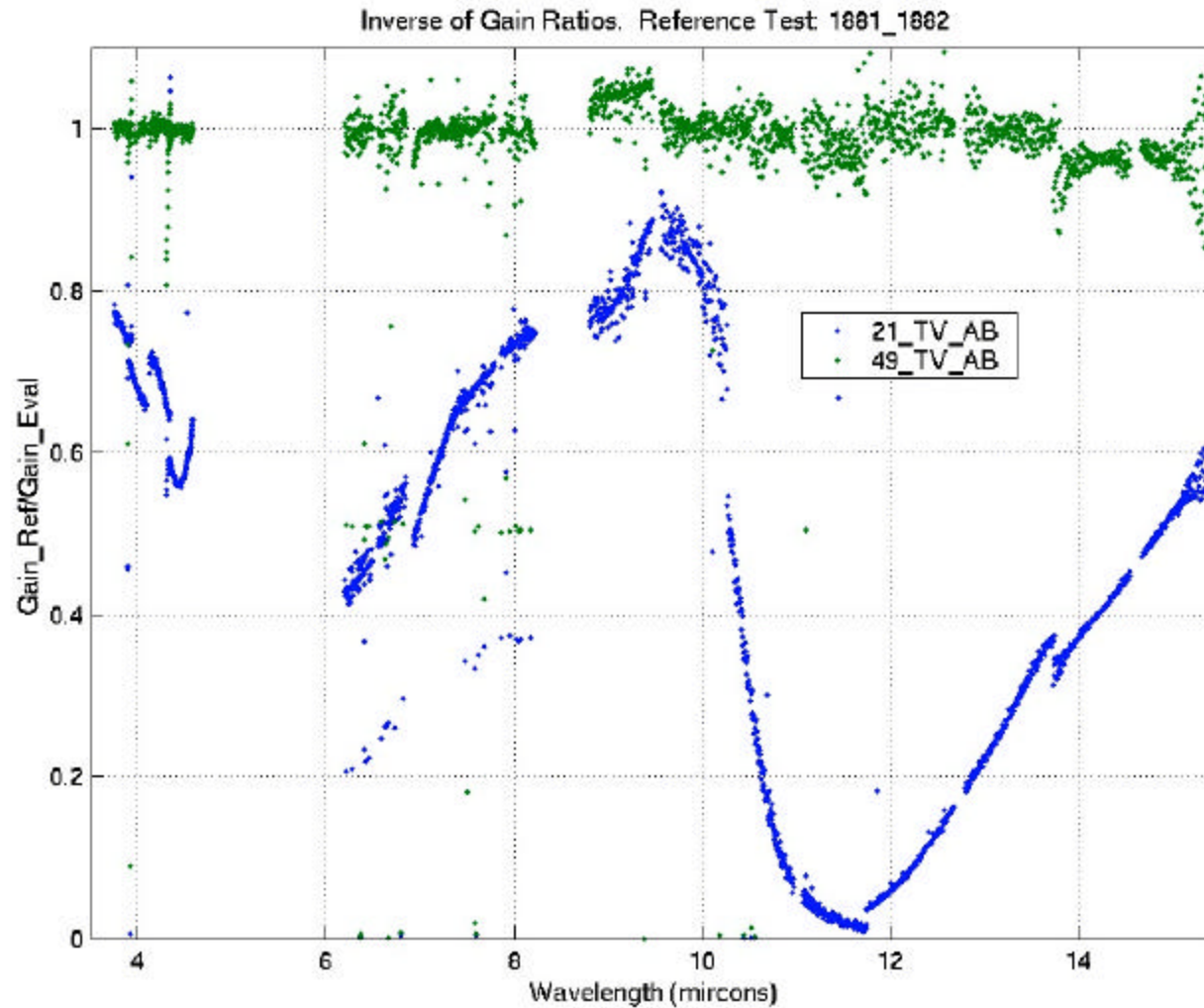
C1: PREDICTED SATURATION LEVELS COMPUTED BASED ON A/D SATURATION



**M1 AND M2
DETECTORS MAY
SATURATE PRIOR
TO A/D SATURATION**



C2: FIRST OBSERVATIONS OF GAINS IN TVAC SHOWED ICING OF OPTICS



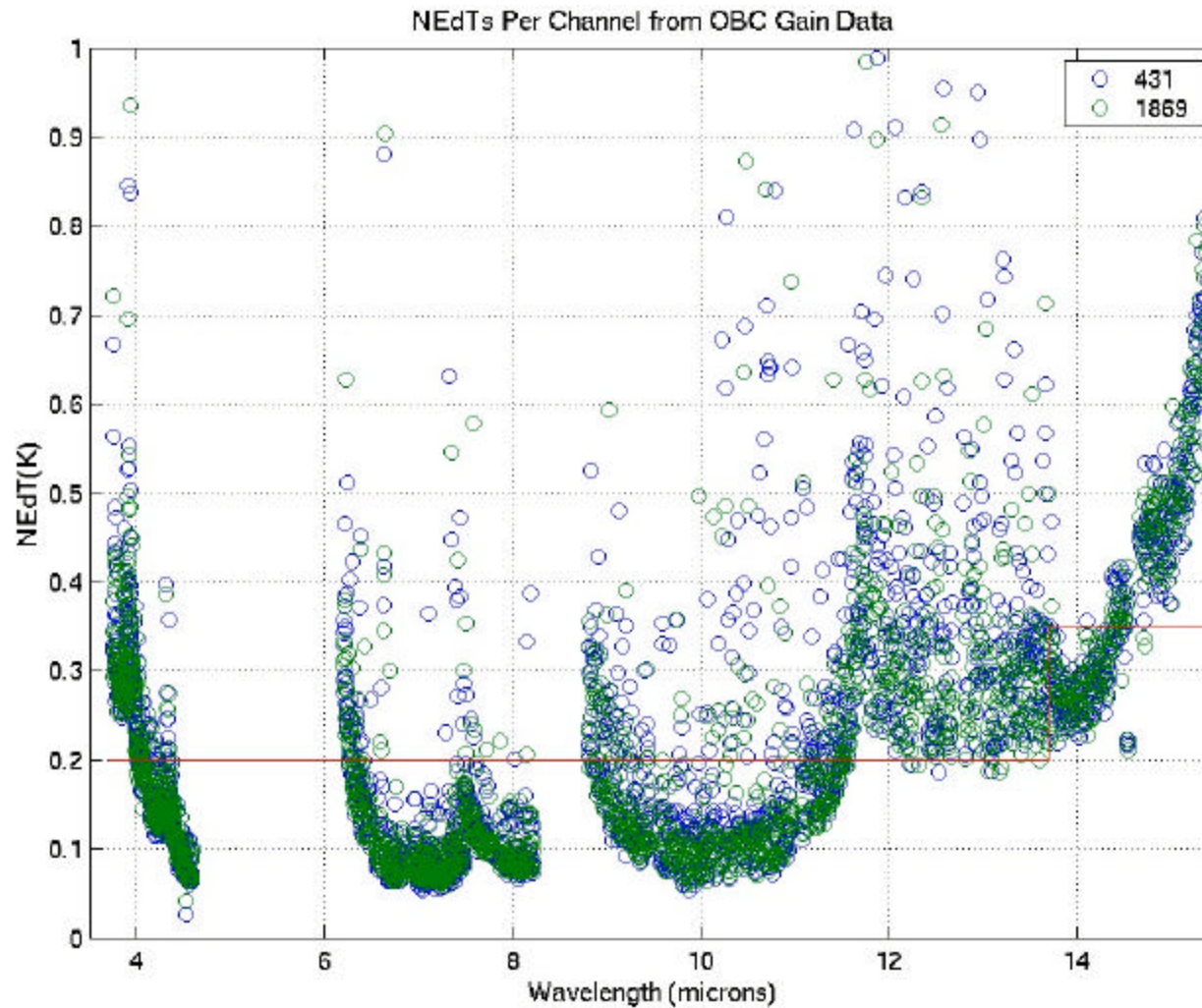
- ICE GONE AFTER OUTGASSING

- REQUIRED ELEVATED TEMP OPERATION UNTIL ICE DISSIPATED

- ICE TRANSMISSION SPECTRUM

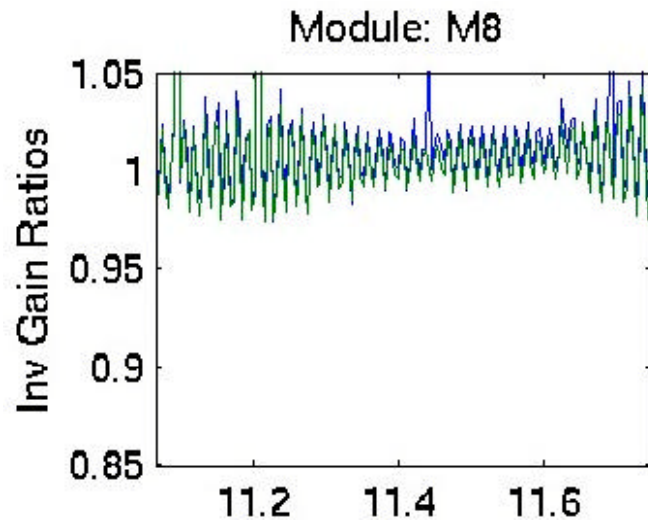


C2: NEDTs COMPARABLE WITH THOSE TAKEN AT BAE SYSTEMS

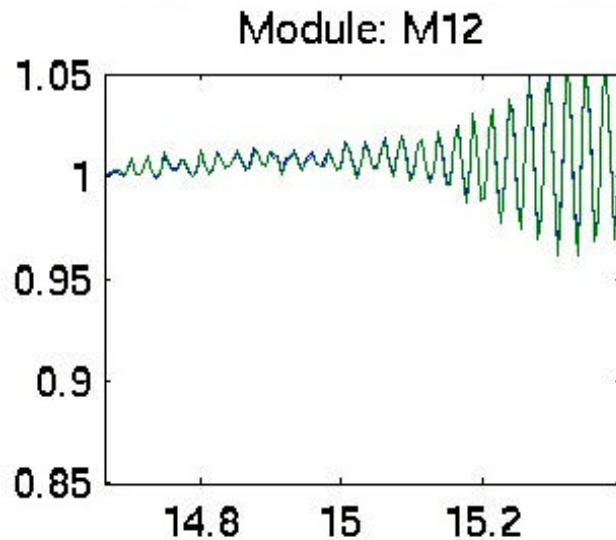




C3: EXCELLENT GAIN STABILITY AND FIDELITY ALLOW CHANNEL SPECTRA PHASE DETERMINATION WORK IN PROGRESS



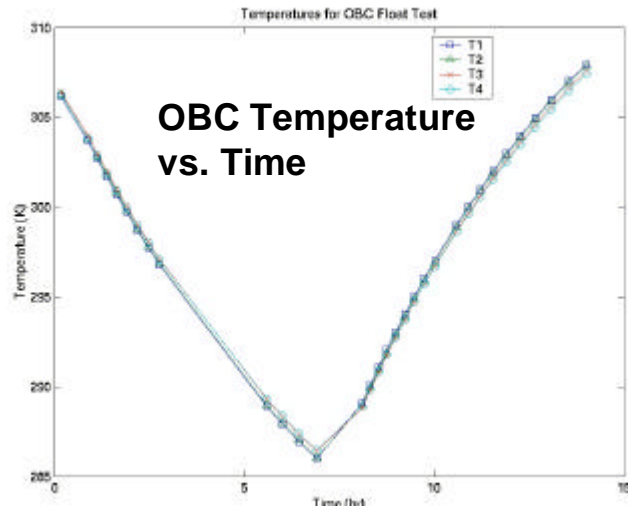
Oscillations due to channel
spectra phase
change with optics
temperature (appx 3° here)



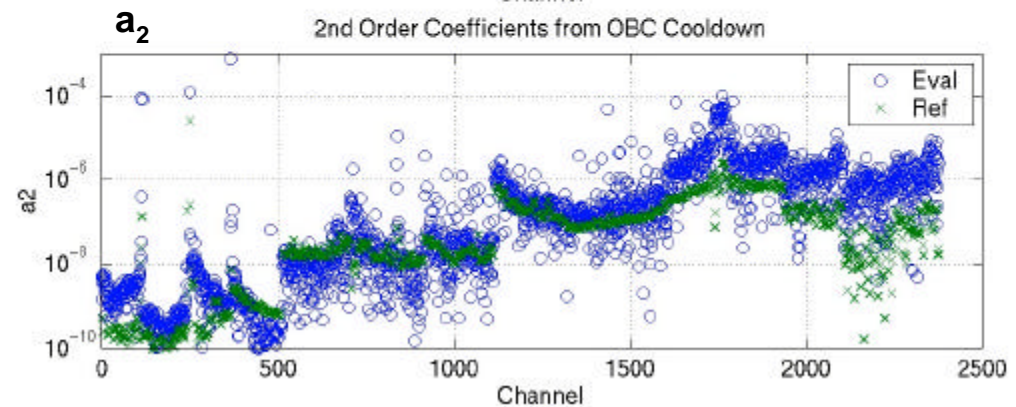
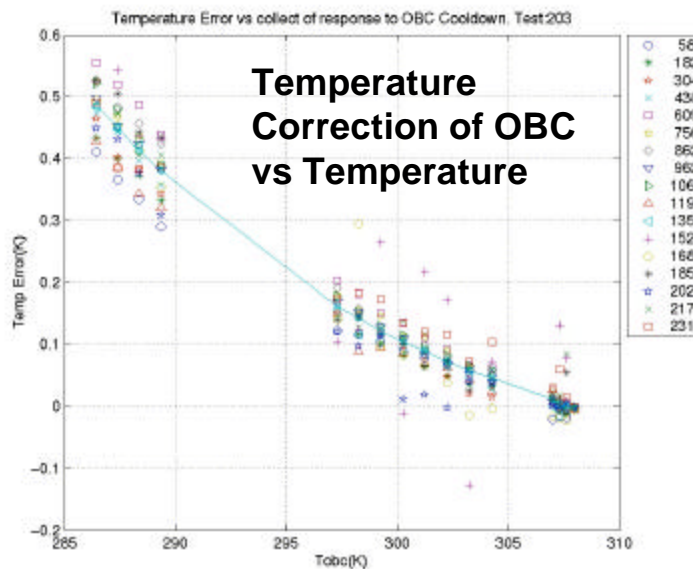
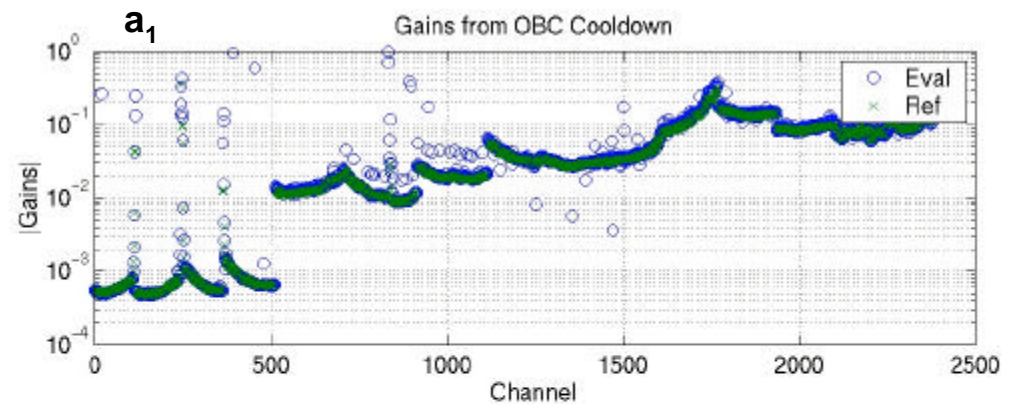
— 78_TV_AB
— 74_TV_AB



C5: OBC FLOAT TEST PROVES ABILITY TO MEASURE ON-ORBIT NONLINEARITY



OBC FLOAT TEST CONFIRMS RADIOMETRIC CALIBRATION COEFFICIENTS ON ORBIT



C6: USES VARIABLE INTEGRATION TIME AND ALSO CHECKS NONLINEARITY

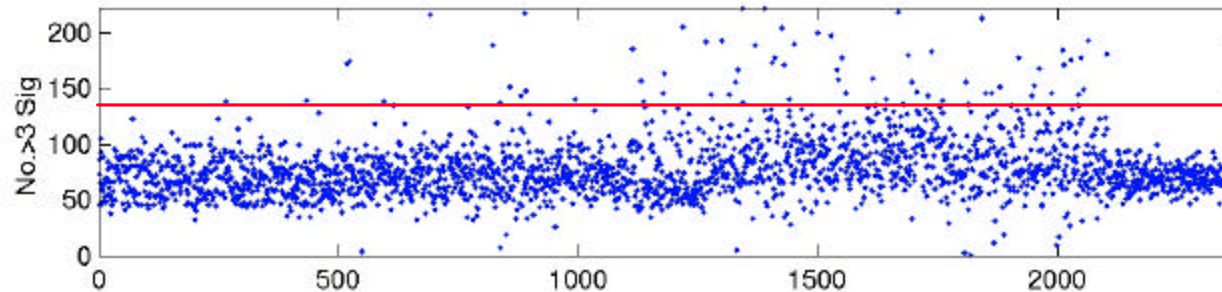
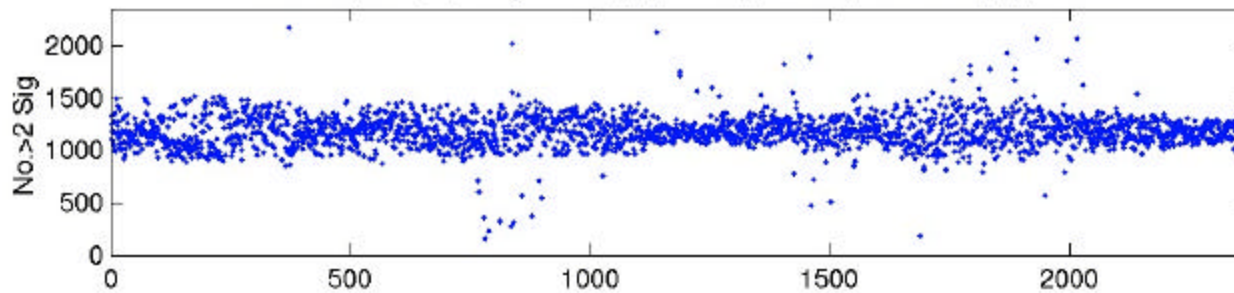


C7: SHOWS MOST DETECTORS HAVE GAUSSIAN NOISE

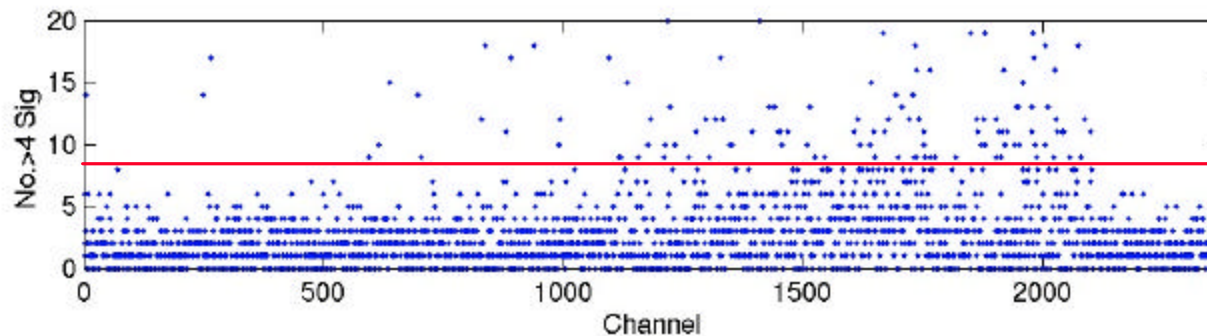


STARE AT SPACE OR OBC AND COLLECT NOISE SAMPLES

No. Samps > 2,3, 4 Sigma: C7_sv_nse/sv_nse/outputs/tv/130_sv_nse.txt



140



8

**NO FIXED
PATTERN NOISE
OBSERVED**

**>25,000 NOISE
SAMPLES
ACQUIRED**

**NUMBER OF
EVENTS > 2, 3, 4
SIGMA COUNTED**

**THRESHOLDS
DEFINED TO
SELECT A AND B
DETECTORS**

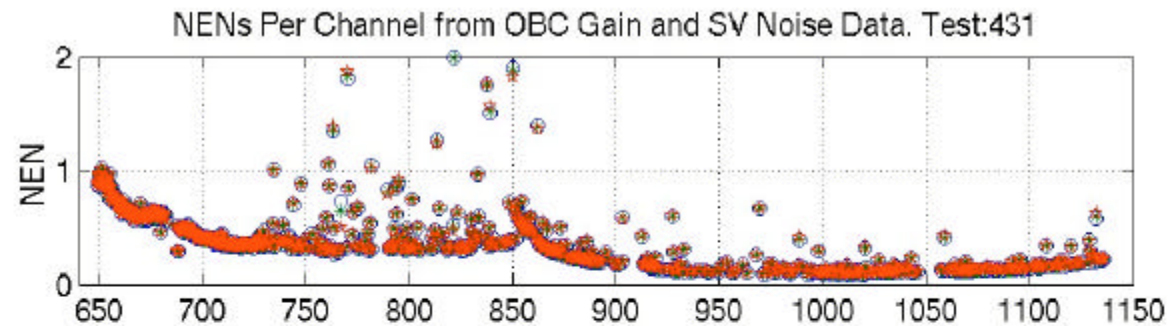


C7: SOME SCENE DEPENDENCE OF NENS FOR M1 AND M2

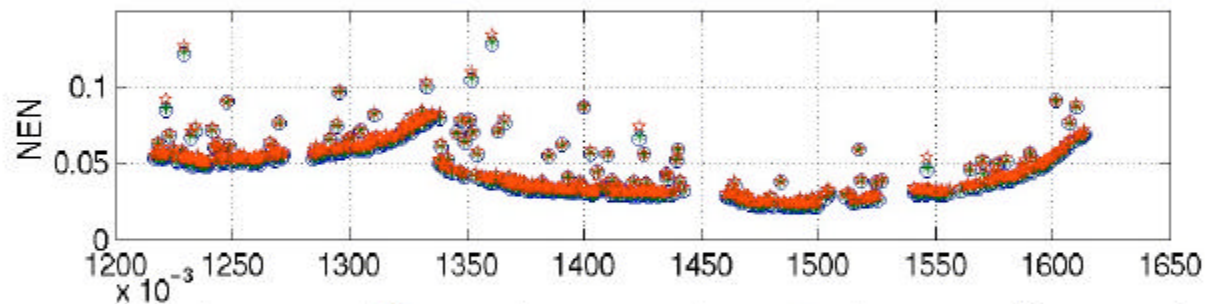


Noise data acquired staring at OBC and SV independently give signal dependence on noise

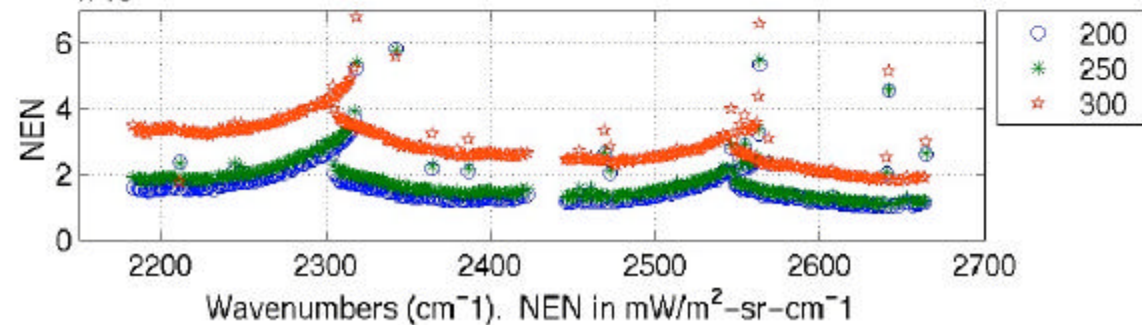
Detector
Noise
Limited
M5-M12



Detector
Noise
Limited
M3, M4



Photon
Noise
Limited
M1, M2



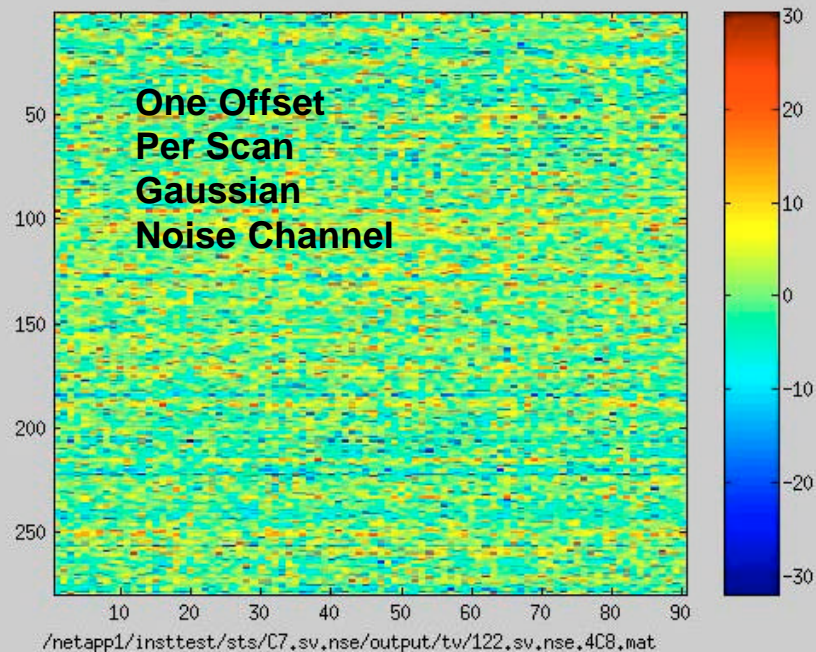


SV SUBTRACTION TECHNIQUE AFFECTS SCENE CORRELATION ERROR

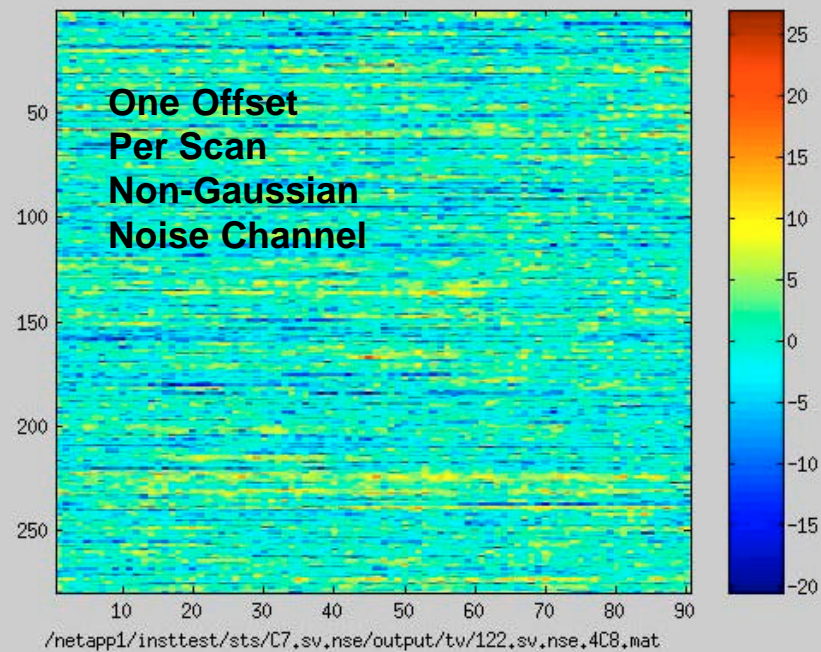


- **At-launch technique**
 - *Subtracts median of 8 space views (offset) per scan line from all footprints in the scan line.*
 - *A new offset is calculated for each scan line*
 - *Results in more noise along track than along scan because all footprints use same space view offset*
- **Alternate technique**
 - *Offset calculated as fit to offset for all scans in the granule*
 - *All scans share a common space view functional dependence*
 - *Results in lower noise correlation error for well behaved detectors*
 - *Expect difficulties when we have DCR or moon in the viewport*
 - *Increases the noise for channels with higher $1/f$ noise*

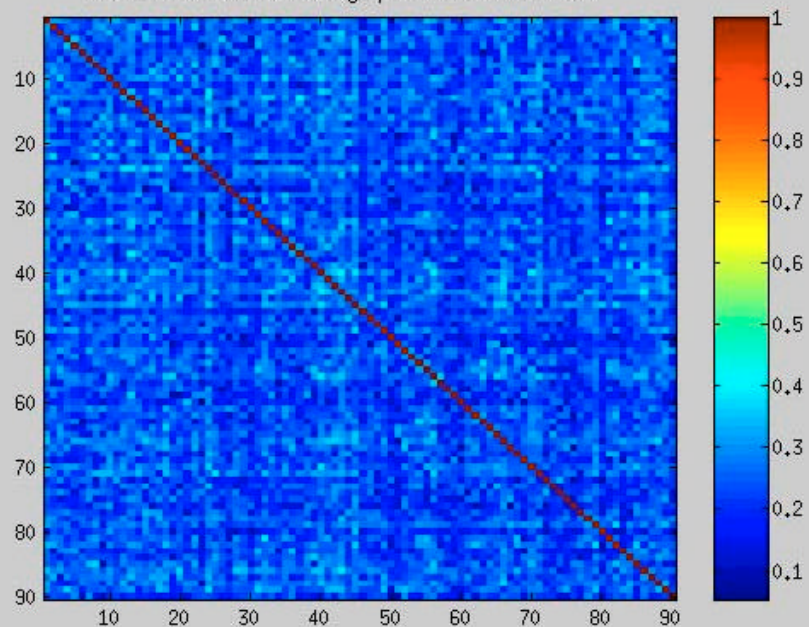
det#2312 std=7.29 Median corrcoeff=0.2424



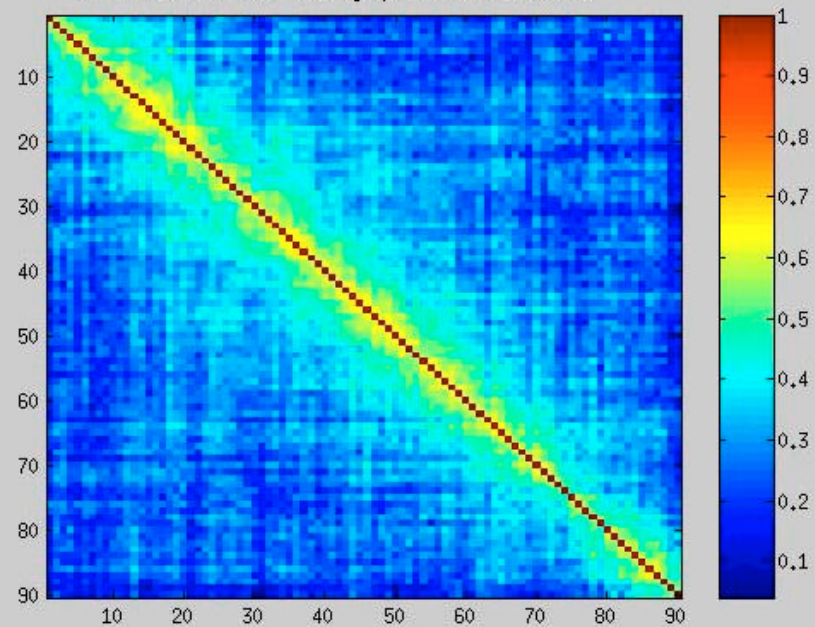
det#2023 std=4.822 Median corrcoeff=0.299



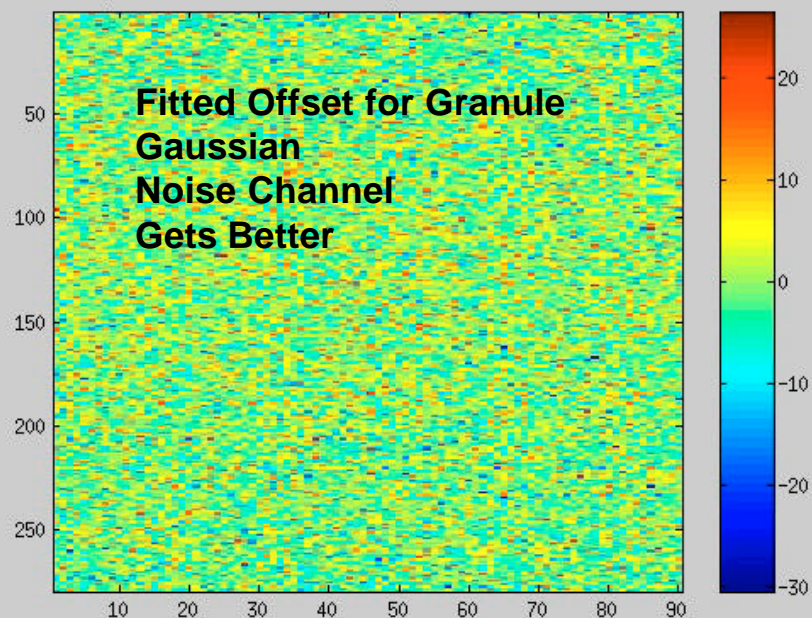
corrcoeff(data) Note the high positive correlation!



corrcoeff(data) Note the high positive correlation!

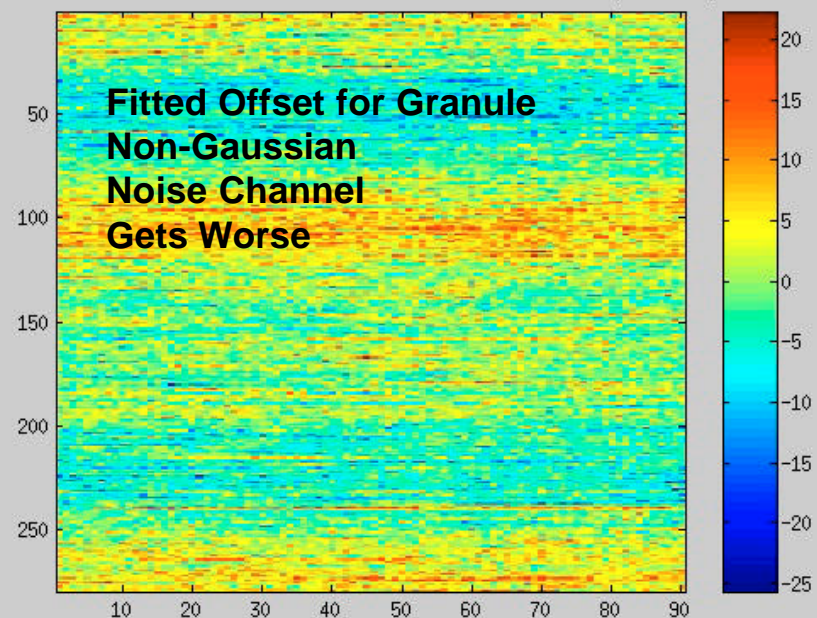


C7,122 SV fitted level1b std=6,38 det#2312 (M12 15um)



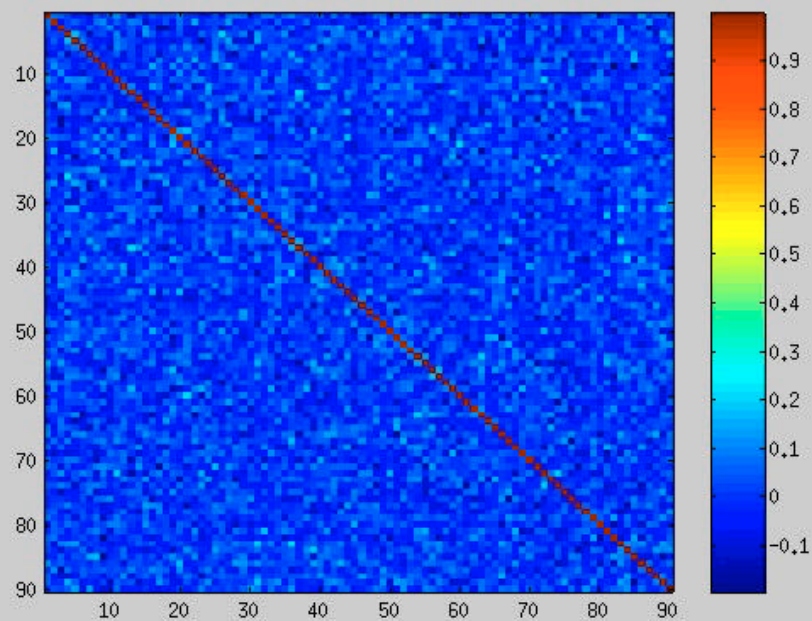
Passed all tests. SV fit processing

C7,122 SV fitted std=5,58 det#2023 (M10 10um) SV fit processing 31oct01

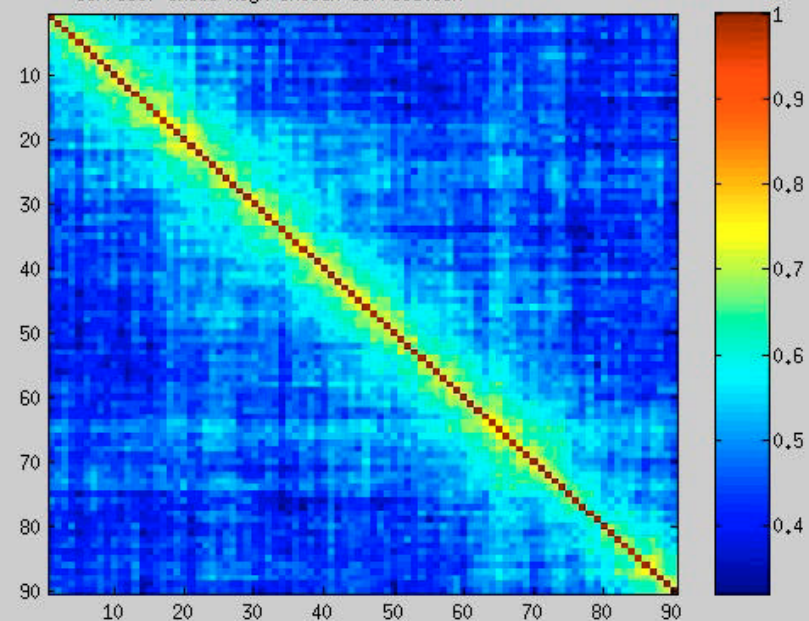


A-side passed all test, B-side failed 3,sig test, passed all 4,sig and 4-4pop tests

Corrcoef textbook for gaussian noise detector



Corrcoef shows high inscan correlation





C8, C9, C10 SPECIAL TESTS RESULTS



- **C8: Radiation Circumvention Test**
 - *Acquired thresholds for one channel per module*
 - *Should be adequate to set levels for all channels*
 - *Requires final setting in orbit in radiation environment*
- **C9: Scan Profile Test**
 - *Rotated scan profile allows us to measure stray light on on-board calibrators*
 - *Test run successfully and data show no anomalies*
- **C10: Vis/NIR Lamps**
 - *Lamps worked well. Vis/NIR responses as expected*
 - *Longer than expected turn on transients require longer wait time after lamp turn on prior to calibration*



SUMMARY AND CONCLUSIONS



- The AIRS instrument performed exceptionally well in T/V
- No problems encountered with AIRS instrument
- We learned a tremendous amount about the instrument
 - *Temperature stability requirements*
 - *Noise behavior*
 - *Alignment methods (AMA)*
 - *Spectral and Radiometric Sensitivity*
 - *Techniques for characterization of performance in orbit*
 - *Some test procedures need modification*
- Special Test Procedures will be performed again during A&E phase and will allow traceability of performance from pre-flight to in-orbit